



# **Application of SpaceWire to the Unionics Advanced On-Board Distributed Processing Architecture**

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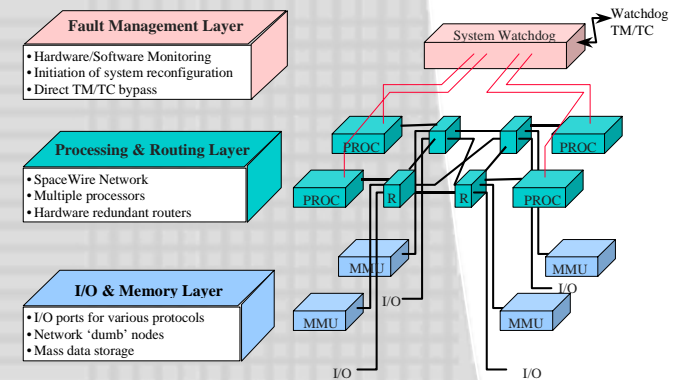
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## The Unionics Project - Objectives

BNSC initiative to develop a high-performance, high-reliability, cost-constrained spacecraft avionics architecture

- Exploitation of existing UK capability and heritage, led by EADS Astrium Ltd (system) with SEA Ltd (hardware) and SciSys (software)
- Development of a robust, scalable processing architecture which is adaptable to a wide range of missions – and has now been baselined!
- Exploitation of space-qualified high-powered processors and SpaceWire standard
- Reusable hardware and software modules allowing scalability.
- Independence from a specific choice of processor.
- Optimised approach to redundancy and fault tolerance
- Open architecture with SpaceWire links that can host third party software and hardware
- Demonstration of an end to end system, one step from a flight system

## Unionics – Main Features

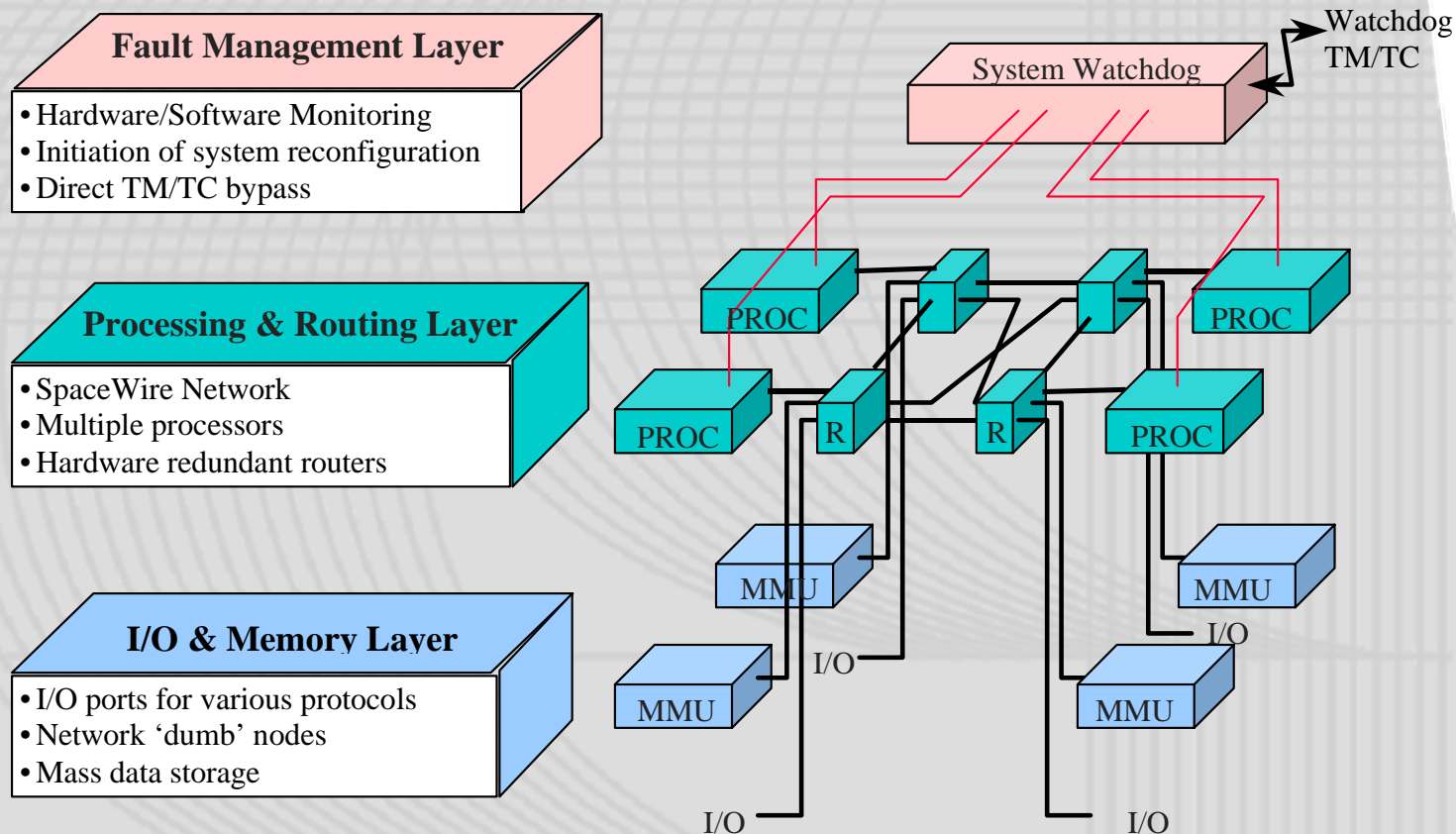
- Increased on-board processing capability (e.g. for instruments with high data rates; autonomous behaviour; formation flying)
- Combination of high throughput payload processing and real-time avionics functions within same system
- Robust fault management approach, with high tolerance to failures of individual elements, despite increasing functionality and complexity
  - Simple, robust watchdog to monitor processor health
  - Transparent relocation of s/w applications following processor failure
- Full resource sharing for resilience.
  - Elegant degradation under failure conditions of total processing resource
  - All peripherals accessible to all processor nodes
- High speed network using Spacewire Router and links
  - SpaceWire compatibility tested with 4 Links router
  - Redundant links provide fault tolerance, relocation of processes and sharing of processing resource



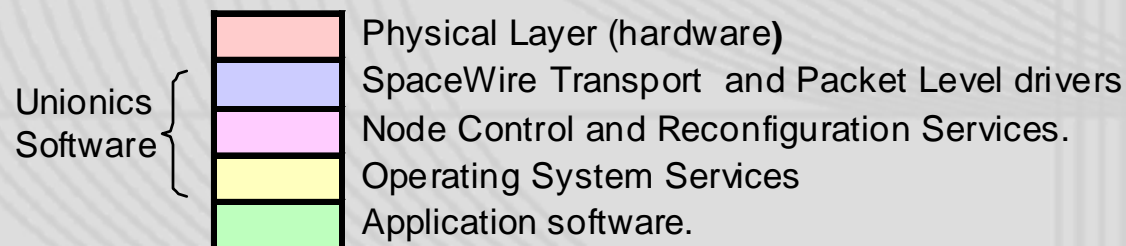
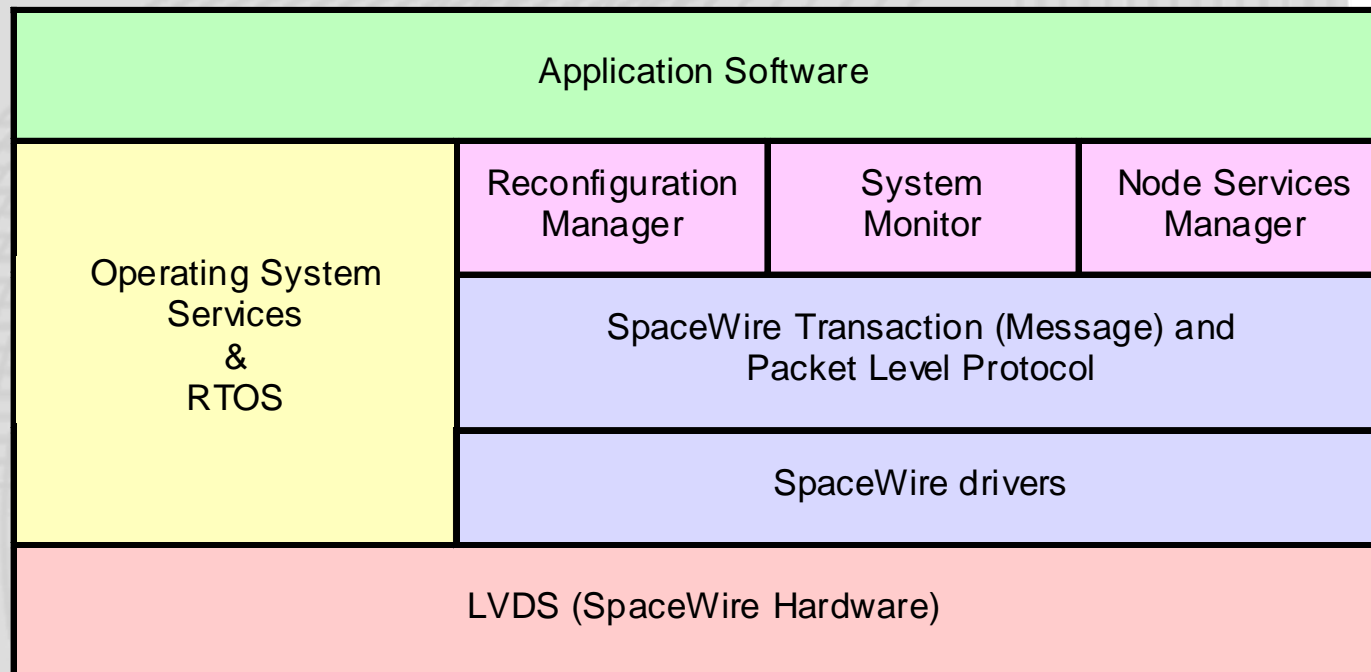
## Layered FDIR and Watchdog

- **Unionics FDIR** capability designed for integration into modern, layered spacecraft FDIR:
  - Lowest level - hot swapping individual units (fail operational).
  - Second level - internal Unionics reconfigurations, (fail operational).
  - Third level - spacecraft level reconfigurations with transition to safe mode (fail safe).
- **System monitor** runs on each processor to check health
  - Runs local reconfiguration manager if a process fails
  - Provides periodic messages from processor node to hardware watchdog
- **System watchdog** monitors health of each processor
  - Generates reconfiguration messages (asynchronous, over LVDS) if message indicates error or if message not received.
- **Reconfiguration manager** is started on one healthy node, which re-starts failed applications on remaining healthy nodes

# Unionics Architecture



# Unionics Software Architecture



## Unionics Software

The Unionics software forms part of the Unionics architecture:

- Real Time Operating System and Services
  - Runs standard ANSI C applications supporting third party developments, legacy code re-use and auto-coding
  - prioritised pre-emptive multitasking
- Spacewire Drivers
- Failure Detection Software Components
  - Report processor and applications status to Watchdog
- Failure Correction
  - Locally restart failed applications
  - Reconfigure Unionics domain according to pre-defined configuration tables in response to Watchdog
- Time Synchronisation
- Processor Mode Services
  - Operating System level communications between processors



## Spacewire Drivers

- Bi-directional communications using Virtual Channels
- Spacewire message consists of a block of data with an added message header and a message header containing the length of the data and its sequence number
- Router design ensures that packets arrive in correct order. Missing packet causes the whole message to be dropped and the calling application to be informed
- Priority has been added to driver to allow use with mixed control and data driven systems (eg AOCS and payload processing)

# Unionics Bench Top Demonstrator

## Objectives of Unionics Demonstrator

The Unionics demonstrator fulfils the following objectives:

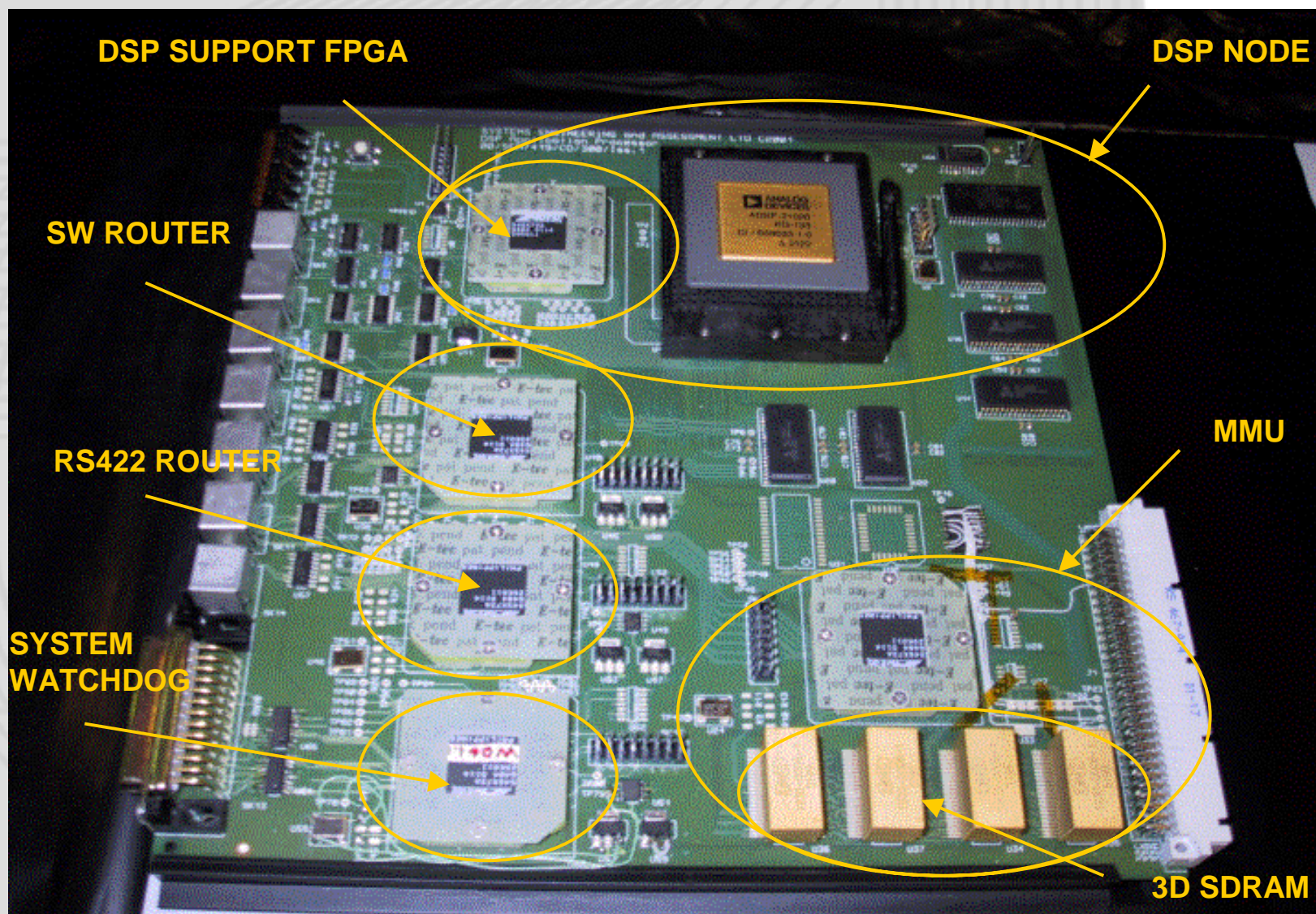
- One step from flight prototype of Unionics hardware and software.
- Test principal Unionics capabilities.
- Test bed for different Unionics strategies, e.g. alternate failure detection & correction implementations.
- Demonstrate Unionics applicability to a range of different missions in a closed loop environment.
- Test bed for prototyping new tools and methods for low cost software application development, e.g. auto-coding
- To publicise Unionics capabilities.
- Early testing of software applications during the development of a specific mission based on Unionics.

## Unionics Demonstrator Scenario

- Three LEO spacecraft, which can operate independently or can utilise inter-satellite links for master/slave communications
  - Validates Unionics applicability to low cost LEO spacecraft, as well as some aspects of trans-spacecraft processing network
- Incorporates AOCS, data handling and payload processing applications.
  - Auto-coding used in AOCS development
  - Existing code reused for payload processing
- Unionics failure detection and correction capability integrated into spacecraft failure detection and correction scheme
- Test environment provides command and monitoring, closed loop dynamics simulator, simulated external units and graphical displays.

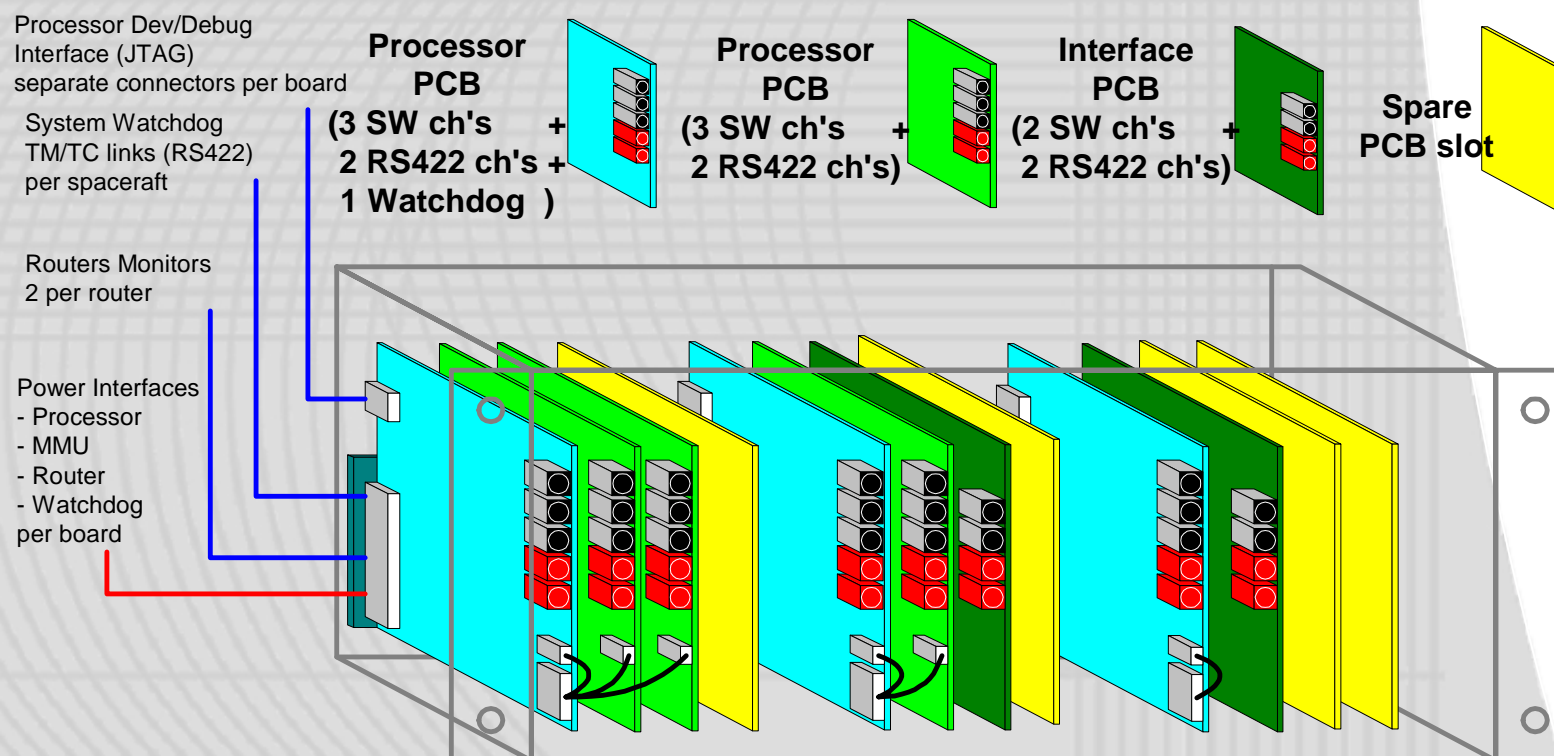


# Unionics Demonstrator Card Layout





# Rack Configuration

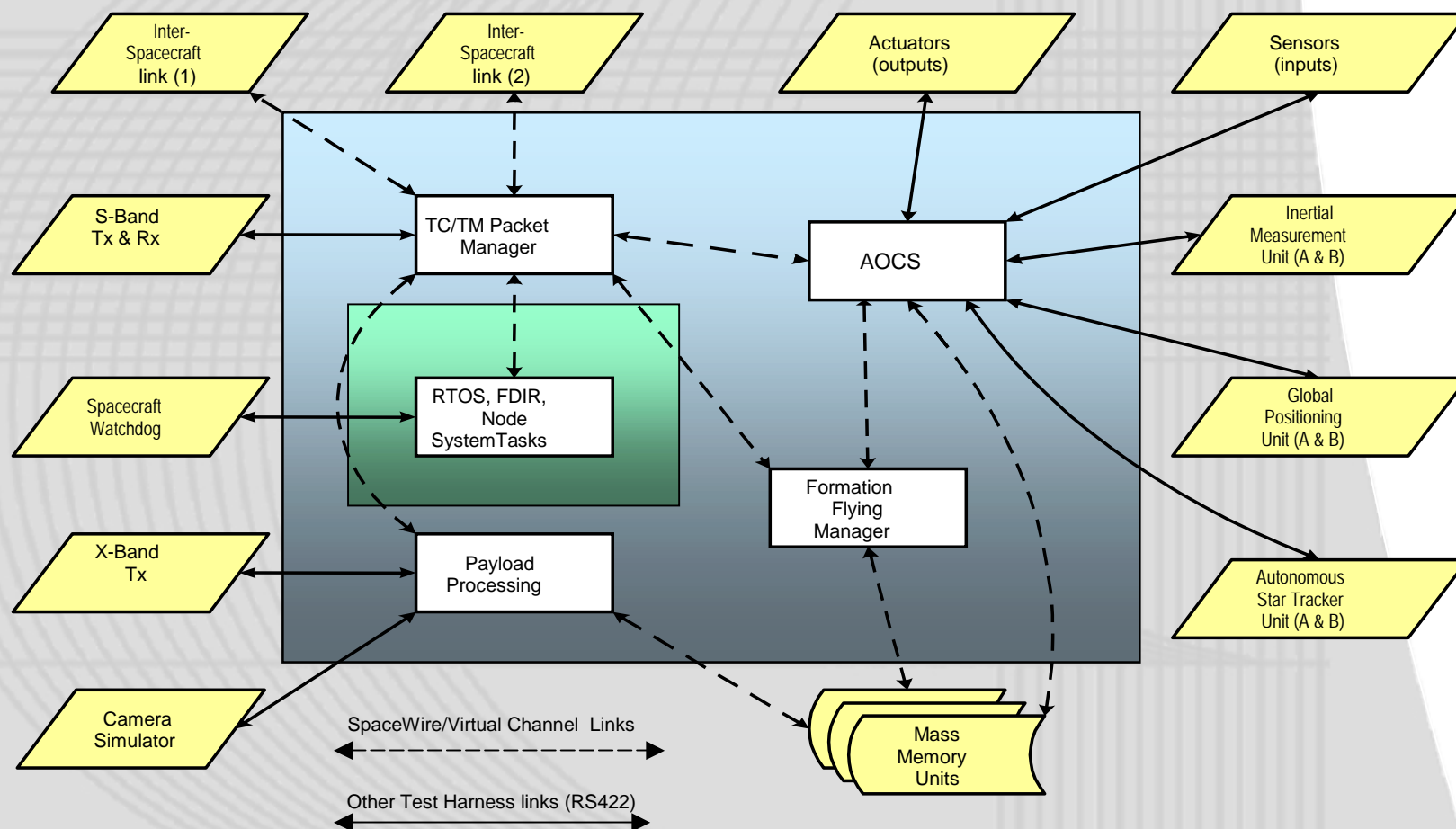


# Unionics and Power Supply

- The Unionics demonstrator rack represents 3 spacecraft
  - S/C #1 – 3 nodes
  - S/C #2 – 2 nodes
  - S/C #3 – 1 nodes
- The power supply unit allows interruption of power to any unit to simulate failures



# Unionics Demonstrator Software



# Unionics Demonstrator Environment

## Real Time Simulator:

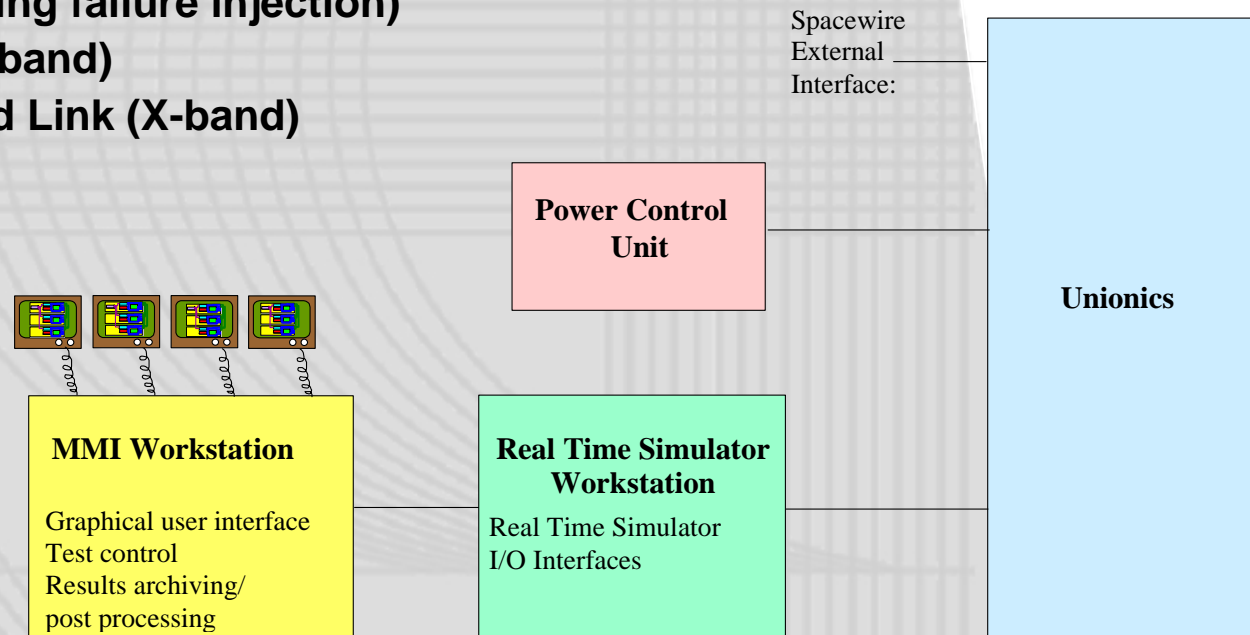
- Unionics I/O
- Spacecraft Dynamics
- On-board Units (including failure injection)
- Ground TM/TC Link (S-band)
- Payload Data Download Link (X-band)
- Payload Observations

Manual control of individual Unionics power supplies

Graphical User Interface

Autonomous test scripts and manual control

Results archiving and post processing

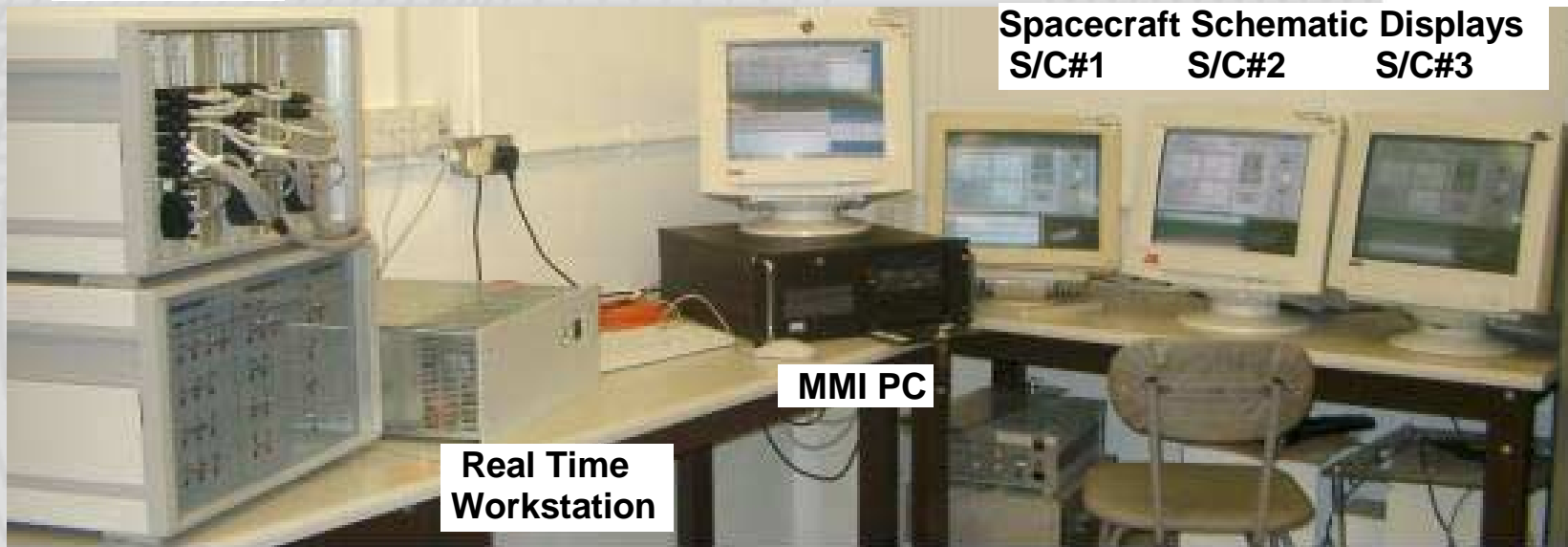


## Demonstrator Hardware

Unionics

Command Screen

Spacecraft Schematic Displays  
S/C#1      S/C#2      S/C#3



Power Supply

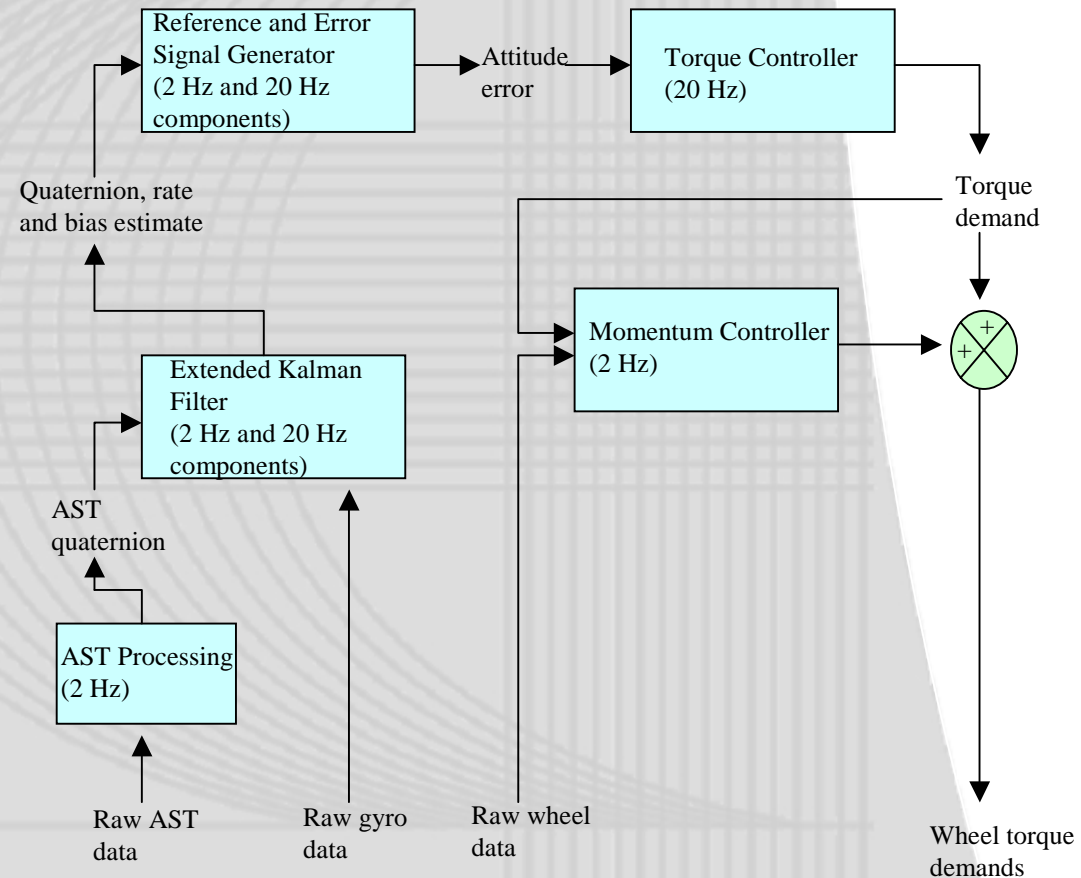


## Closed-loop real-time AOCS

- AOCS and dynamics simulator developed and tested in Simulink.
- AOCS autocoded using Real Time Workshop (RTW), compiled in Microsoft Visual C++, and validated against Simulink model.
- Autocode compiled/linked with handcoded AOCS functions (I/O to sensors/actuators etc) using visual DSP. Loaded onto Unionics
- Simulink model dynamics simulator autocoded, and integrated onto test set real time workstation (Dspace).
- AOCS on Unionics validated closed loop with dynamics simulator on test set.

# AOCS Architecture

- Represents normal mode utilising Autonomous Star Tracker (AST), Gyro and wheels.
- State estimation utilising extended Kalman filter
- 10 Hz control law frequency, with 2 Hz updates for AST and wheels



# **Introduction to the Unionics Demonstration**

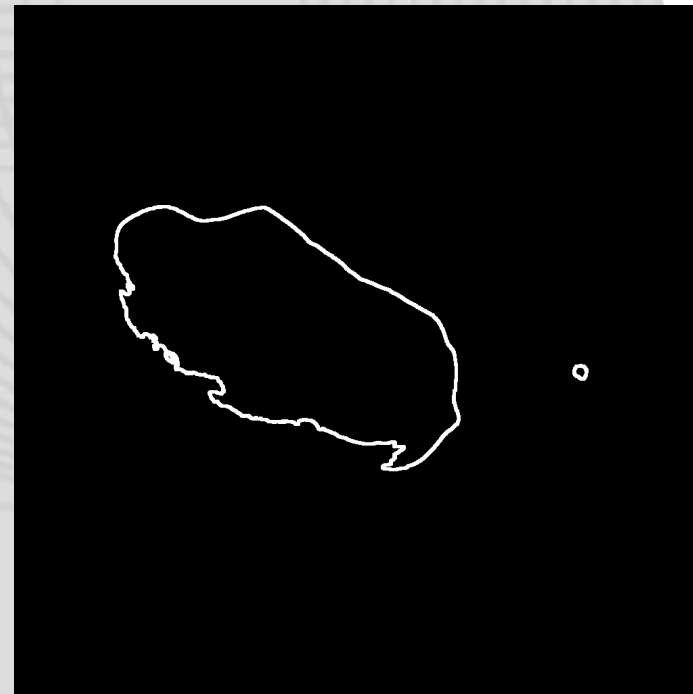
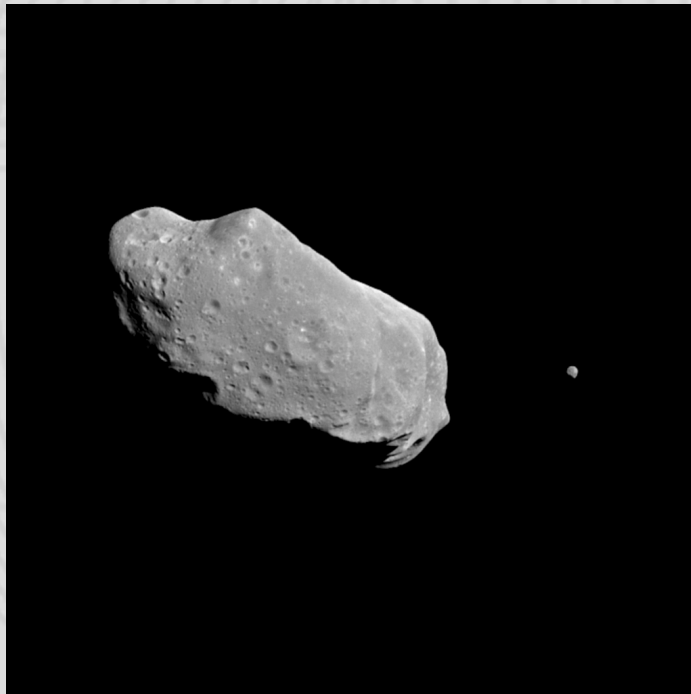
## Unionics Demonstration (1)

Three key features of the Unionics demonstrator will be presented:

- Simultaneous operations of data handling, simplified AOCS and payload processing, on a spacewire network based distributed processing architecture.
- Trans-spacecraft network employing spacewire inter-satellite links, and a change of the formation master spacecraft
- Unionics inherent FDIR automatically detecting processor failures and reconfiguring processes to processors mapping.

## Unionics Demonstration (2)

- Simultaneous payload processing
  - Reads in an asteroid image and computes its outline, and returns the processed image.





# Graceful Degradation

